

(12) **United States Patent**
Fouche(10) **Patent No.:** **US 6,751,529 B1**
(45) **Date of Patent:** **Jun. 15, 2004**(54) **SYSTEM AND METHOD FOR CONTROLLING MODEL AIRCRAFT**(75) Inventor: **J. Michael Fouche**, Huntsville, AL (US)(73) Assignee: **Neural Robotics, Inc.**, Huntsville, AL (US)

(*) Notice: Subject to any disclaimer, the term of this patent is extended or adjusted under 35 U.S.C. 154(b) by 0 days.

(21) Appl. No.: **10/449,372**(22) Filed: **May 30, 2003****Related U.S. Application Data**

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(51) **Int. Cl.⁷** **B64C 11/34**(52) **U.S. Cl.** **701/3**; 244/3.21; 244/164; 244/171; 342/29; 340/967(58) **Field of Search** 701/3, 4, 7, 48; 244/3.21, 164, 171, 181, 183, 158 R, 177, 179; 342/29, 30; 340/967, 975, 978(56) **References Cited****U.S. PATENT DOCUMENTS**

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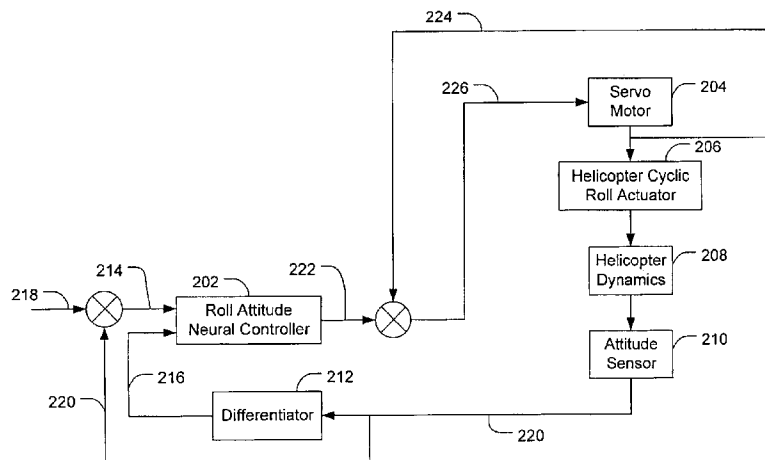
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Primary Examiner—Thomas G Black*Assistant Examiner*—Tuan C To(74) *Attorney, Agent, or Firm*—Lanier Ford Shaver & Payne P.C.(57) **ABSTRACT**

In one embodiment, a method for controlling an aircraft comprises providing an attitude error as a first input into a neural controller and an attitude rate as a second input into the neural controller. The attitude error is calculated from a commanded attitude and a current measured attitude, and the attitude rate is derived from the current measured attitude. The method also comprises processing the first input and the second input to generate a commanded servo actuator rate as an output of the neural controller. The method further comprises generating a commanded actuator position from the commanded servo actuator rate and a current servo position, and inputting the commanded actuator position to a servo motor configured to drive an attitude actuator to the commanded actuator position. The neural controller is developed from a neural network, wherein the neural network is designed without using conventional control laws, and the neural network is trained to eliminate the attitude error.

28 Claims, 16 Drawing Sheets

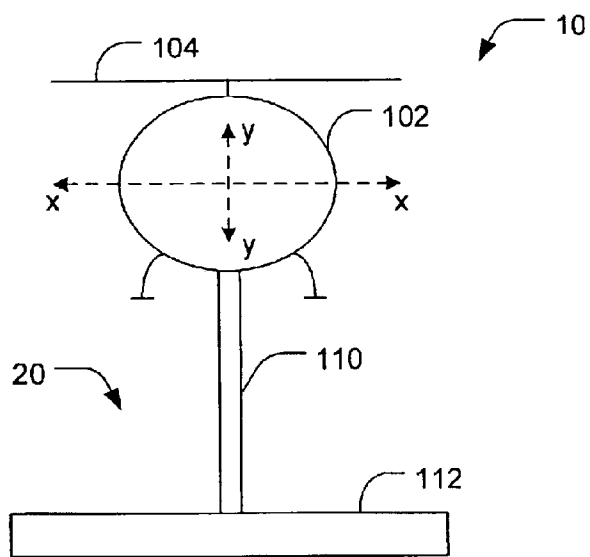
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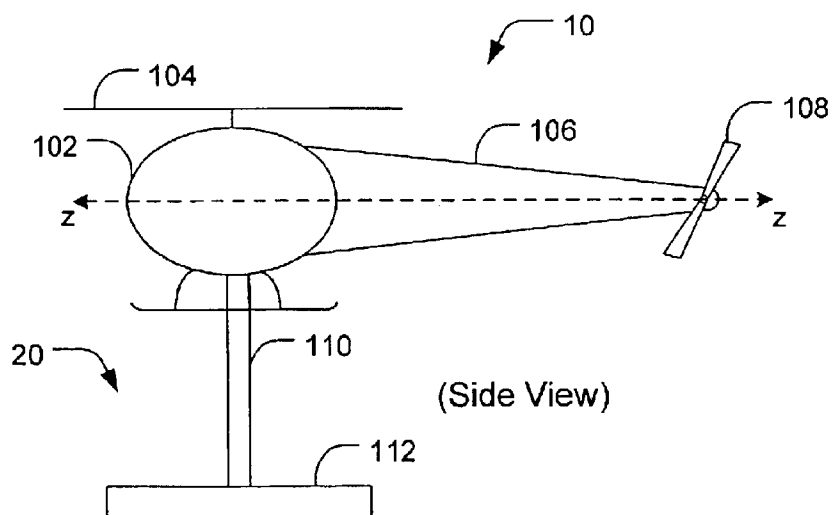
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(Front View)

FIG. 1A

(Side View)

FIG. 1B

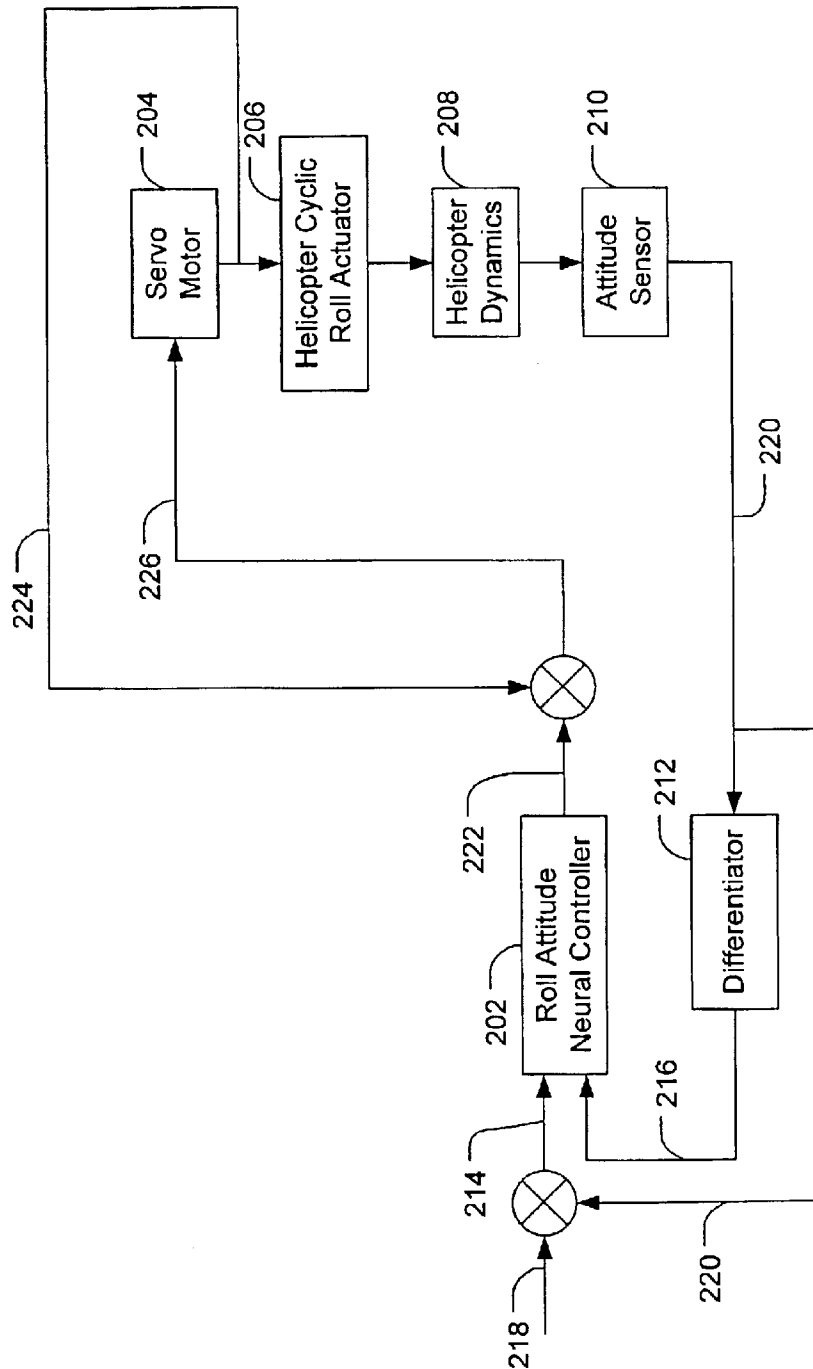


FIG. 2

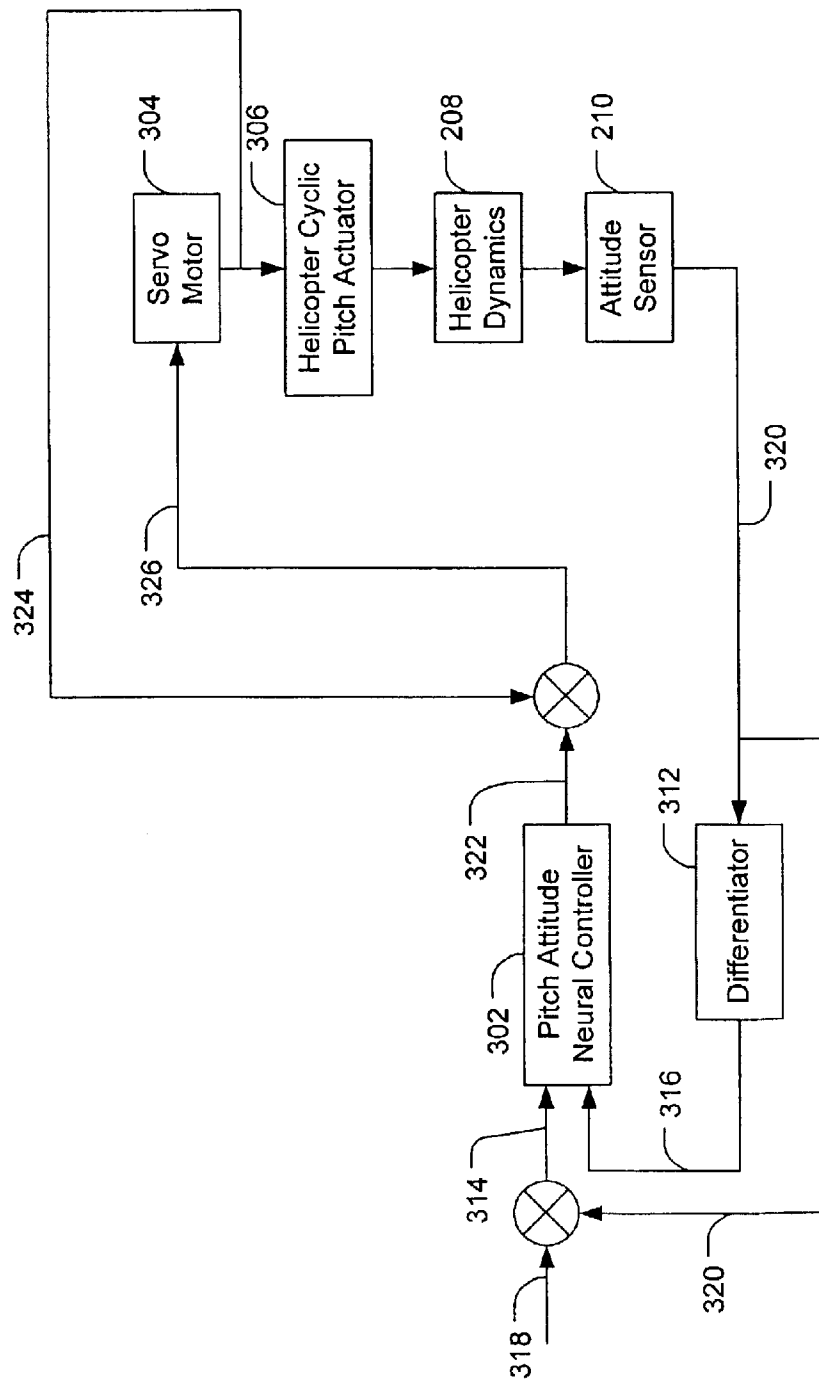


FIG. 3

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